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Let mass of rocket + propellant

$$M = M(t) = M(0) - m(t) \quad \text{s.t.} \quad m(0) = 0$$

$$M(0) = M_0 + m_p \quad m(t) = m_p$$

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$M(t) = M_0 - \dot{m} t$   
(assume constant propellant flow rate),  $M = M(0) - \dot{m} t$   
 $\dot{m} = \frac{dM}{dt}$   
 $M(t) = M(0) - \dot{m} t$   
 $\dot{m} = \frac{dM}{dt}$   
Euler equations (fluid dynamics)", Wikipedia

## Contents

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